NASA Dryden Status

Aerospace Control & Guidance Sub-committee

Meeting 105

Lake Tahoe, NV

March 2010

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IRAC F-18 #853 Testbed

- •! Dedicated Ghz processor for experiment
- ! Shell & process for Simulink autocode (or c-code)
- •! Can control commands to:

All aero surfaces (except speed brake)

All pilot inputs

Both engine throttles independently

- ! Limit checks done by Class A software in RFCS
- •! Potential for Class A experiment (dual ARTS IV or in quad RFCS) take to landing?
- •! Tons of research instrumentation parameters (mostly related to structures)
- •! Simulated A-Matrix and B-Matrix failures





NASA Dryden Flight Research Center Photo Collection http://www.dfrc.nasa.gov/Gallery/Photo/index.html NASA Photo: EC04-0361-16 Date: December 15, 2004 Photo By: Carla Thomas

NASA's flexible-wing F/A-18 maneuvers through a test point during the second phase of the NASA/Air Force Active Aeroelastic Wing flight research program.



IRAC F-18 #853 Testbed - Current Status

•! Completed Hardware-in-the-loop testing (August)

•! First flight of new hardware — March 2010 – ON SCHEDULE

•! Dynamic Inverse controller – June 2010

•! Simplified Model Ref Adaptive Control — August 2010

•! Evaluating simplified adaptive control approaches

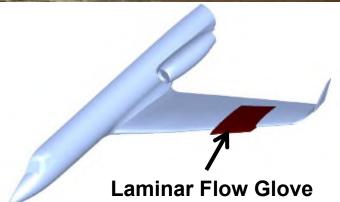
- ! Dynamic Inversion Baseline
- •! Simplified MRAC
- ! Benefit vs. complexity trade studies on extensions to basic MRAC
- •! Interaction between adaptive controller and aircraft structure
 - •! Investigating ways for pilot to control learning rates
 - •! Planning to fly cross-coupling handling qualities metric development test with AFFTC test pilot school
 - •! Future planned work
 - •! Adaptive controller implemented in redundant system



NASA G-III Research Aircraft

- •! NASA DFRC is acquiring a Gulfstream III (G-III) to serve as a flying testbed for aeronautics experiments
- •! The aircraft will be instrumented and modified to accommodate a range of flight testexperiments
- •! Laminar Flow Glove
 - •! NASA's ERA program is funding a flight-test of a wing glove with a natural laminar airflow airfoil. Discrete Roughness Elements (DRE)s will be placed on the glove for passive laminar flow control. Texas A&M and Dryden will be developing the glove.
- •! Adaptive Compliant Trailing Edge (ACTE)
 - •! AFRL is funding development and flight test of an adaptive, compliant flap. The port inboard flap of the G-III will be replaced with a compliant design. The flight test will examine ACTE suitability as a lift control device (flap), control surface (ailerons), and trim device (trim tabs).
- ! Aircraft acquisition planned for early CY 2010.





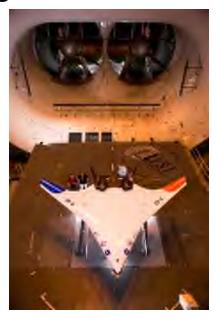


Adaptive Compliant Trailing Edge



X-48 Blended Wing Body

- •! 74 flights completed on X-48B
 - -!Slats extended and slats retracted stall onset has been characterized
 - -!Flight results providing data for aerodynamic model and simulation updates
 - -! Currently flight testing departure limiter assault
- •! Peak seeking control to optimize inflight drag reduction in 2010





- •!X48C completed wing tunnel testing
- !Preparation work on X-48C for flight
 - -FEM, simulation, engine integration, and control law development
 - -Design and build flight weight parts
 - -Complete modifications and prepare for flight

SOFIA

- •! Stratospheric Observatory For Infrared Astronomy
 - -! 2.5 m diameter German built infrared telescope
 - –! Open port cavity
 - »! ~24°-57° viewable elevation range
 - -! Platform is Boeing 747 SP
 - »! Capable of 6+ hours of observation time
- •! On going open door envelope expansion flights through 2010
 - Concurrent with mission system build up and limited science missions
 - -! Completed two open door straight and level flights at 10% and 100% open in December 2009
 - -! Completed two expansion flights at 10% and 40% open up to 15Kft and 225 kias January 2010

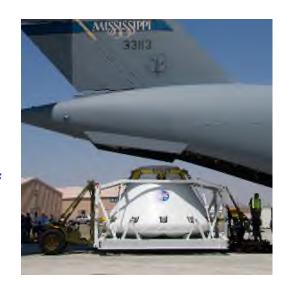


- -! Completed one shortened 70% open expansion flight February 2010
- -! Envelope clearance with a cavity acoustics focus
 - »! All test points show sound pressure levels are below expected levels thus far
- ! Autopilot interface development to support science mission navigation requirements is ongoing



Orion CEV Launch Abort Systems Tests

- ! Dryden is leading the test activities for the Launch abort systems test. Tests will be conducted at White Sands, NM
 - -! Pad Abort 1 (PA-1): Tests the basic functionality of the launch abort system from the pad in its preliminary design configuration.
 - »! Current launch date is late April early May 2010
 - -! Ascent Abort 2 (AA-2): Tests the ability of the launch abort system to function as the spacecraft approaches the region of maximum drag.
- •! Current program status is to continue with the abort flight tests as planned through Sept 2010 (PA-1)
- •! AA-2 is also still currently in the plans as a technology demonstration
- Plans in work for everything after PA-1 test
- ! Current activities
 - -! Hardware testing and integration of the PA-1 crew module at White Sands, NM
 - -! Preparation for Flight Test Readiness Review
 - -! Planning for future flight tests (eg. AA-2)





To Fly What Others Imagine ...!